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ARTICLE

Flow and Heat Transfer Features of Supercritical Pressure CO² in Horizontal Flows under Whole-Wall Heating Conditions

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ABSTRACT

Based on the first and second laws of thermodynamics, the heat transfer and flow (thermohydraulic) characteristics of horizontal supercritical pressure $CO₂$ (S-CO₂) in a circular pipe under heating conditions were investigated numerically. Heating flows in two different diameters (*d*) of 4 and 6 mm were simulated in pipes with pressures of 8 MPa, mass fluxes (*G*) of 300 and 400 kg/(m²·s), and heat fluxes (*q*) of 50, 75 and 100 kW/m². In the $d = 4$ mm pipe, the peak heat transfer coefficient (h_b) was about 3 times higher than in the $d = 6$ mm pipe, while the entropy production due to fluid friction in the 4 mm pipe was on average 1.1 times higher, and the entropy production due to heat transfer was on average about 67% lower. A 4 mm tube was employed to further evaluate the influence of the applied wall heat flux, the results demonstrated that the irreversibility due to heat transfer was on average more than 4 times higher when heat flux density was 100 kW/m² than when the heat flux density was 50 kW/m², while the peak of heat transfer coefficient increased by 1.4 times as q was decreased from 100 to 50 kW/m². The effect of thermal acceleration was ignored, while the buoyancy effect resulted in secondary flow and significantly affected the flow and heat transfer features. The jet flows were found in the vicinity of the lower wall of the pipe, which made the two fields of velocity and temperature gradient more synergistic, leading to an enhancement in heat transfer in the vicinity of the upper wall. The aggravation of heat transfer resulted in high irreversibility of heat transfer in the cross-sectional area near the wall, while the local friction irreversibility was less affected by the buoyancy effect, and the distribution was uniform. The uneven distribution of thermophysical properties also confirmed that the enhanced heat transfer occurred near the wall area at the bottom of the pipe.

KEYWORDS

Effect of buoyancy; heat transfer deterioration; irreversible loss; supercritical pressure $CO₂$ (S-CO₂); field synergy principle

Nomenclature

Abbreviations

 c_p Specific heat, $J/(kg·K)$
d diameter of tube. m diameter of tube, m

Symbols

Subscripts

1 Introduction

In recent decades, clean and efficient energy conversion technologies have become increasingly attractive as a way to reduce excessive consumption of fossil fuels and greenhouse gas emissions in response to the energy crisis and climate change. Supercritical CO₂ Brayton cycle system is recognized as one of the most promising energy conversion technologies in various fields such as solar energy and geothermal energy [\[1,](#page-18-0)[2\]](#page-18-1). In addition, carbon dioxide is already used in various energy systems because it is chemically stable, non-flammable, and non-toxic [\[3](#page-18-2)[–6\]](#page-18-3). However, the drastically varying thermophysical properties of S-CO₂ and the associated complex thermohydraulic characteristics pose serious challenges to $S-CO₂$ systems and the related equipment $[7–10]$ $[7–10]$. Therefore, the study of the thermohydraulic properties of S-CO₂ under different conditions has aroused widespread interest in order to capture complex features and promote relevant basic understanding $[11-13]$ $[11-13]$.

The characteristics of heat transfer and flow in vertical $S-CO₂$ flows are of interest because the flow direction can be the same or opposite to the buoyancy. Bruch et al. [\[14\]](#page-18-8) experimentally studied the heat transfer and flow characteristics of $S-CO₂$ in vertical pipes under cooling conditions, and reported that the heat transfer coefficient increased with the increase of mass flow in upward flow, and increased with the decrease of mass flow in downward flow after mass flow reached a threshold. Kim et al. [\[15,](#page-18-9)[16\]](#page-18-10) conducted an experimental study on the thermohydraulic characteristics of vertical S-CO₂ flow under heating conditions, and the results showed that the temperature of wall raised monotonically in the flow direction for downward flow, while it showed an obvious peak value during upward flow. Lei et al. $[17]$ conducted experimental research on the heat transfer characteristics of vertical S-CO₂. flow under different heat-flux and mass-flux conditions. The results showed that the buoyancy effect had a key influence on the heat transfer performance, and the difference was significant when the heat flux varied between 38 and 234 kW/m². It has also been reported that under low heat flux conditions, the increase in pressure led to a significant decrease in the heat transfer coefficient, while when the heat flux was high, the effect of pressure was not significant because the fluid temperature in the near-wall region was far away from the pseudo-critical temperature (T_{pc}) . Guo et al. [\[18\]](#page-19-0) conducted a numerical study on the heat transfer and flow characteristics of $S-CO₂$ flow in a vertical pipeline. The results showed that under cooling conditions, the buoyancy effect enhanced heat transfer in upward flow, but impeded heat transfer in downward flow, and the effect of buoyancy effect was opposite under heating conditions. Wang et al. [\[19\]](#page-19-1) carried out numerical studies on the flow and heat exchanger features of $S-CO₂$ vertically upward flow with a lattice structure array, with results showing that the lattice structure could suppress the heat transfer deterioration and reduce the wall temperature peak, since vortexes generated by lattice structure enhanced turbulent and weakened the buoyancy effect and flow acceleration.

Horizontal S-CO₂ flows are more ubiquitous in practical applications. Lin et al. [\[20\]](#page-19-2) summarised existing heat transfer correlations for S-CO₂ horizontal flows at cooling conditions, and reported that most correlations had a good agreement with experimental data at low heat-flux conditions, while the correlations were not applicable when the heat flux was high as the buoyancy effect became significant. Oh et al. $[21]$ performed experimental studies on the heat transfer performance of S-CO₂ in horizontal pipes with inner diameters of 4.6 and 7.8 mm under cooling conditions, the results showed that the increasing pressure resulted in a decrease in the heat transfer coefficient, and the heat transfer coefficient further decreased as the mass flux increased. Liu et al. [\[22\]](#page-19-4) experimentally studied the heat exchange performance of $S-CO₂$ in horizontal pipes with inner diameters of 4, 6 and 10.7 mm under the condition that fluid was cooled, and they reported that the pipe diameter affected the heat exchange performance significantly. Moreover, the correlations developed for small pipes presented notable deviations from the experimental data for large pipes, and hence a new correlation was proposed for large-diameter pipes. The thermal-hydraulic characteristics of $S-CO₂$ in a non-uniform heating horizontal tube was experimentally investigated based on pseudo-critical theory in [\[23\]](#page-19-5), the experimental results demonstrated that heat transfer deterioration was more prone to appear near the pseudo-critical pressure at smaller heat-flux condition, and horizontal flow under non-uniform heating exhibited similar heat transfer characteristics to that in vertical tubes.

To further enhance the heat transfer performance, a modified aerofoil fin was proposed, which enabled an increase of 6% in heat transfer factor and a decrease of 4% in pressure drop relative to the NACA 0020 aerofoil fin [\[24\]](#page-19-6). Han et al. [\[25\]](#page-19-7) further demonstrated that the design of front-dense and rear-sparse fins could intensify heat transfer by 23%–29% as compared to the uniform fin distribution. Altering the cross-section of heat exchange tube was also employed to enhance heat transfer of $S-CO₂$, with resulting showing that the converging tube exceeded the diverging tube in terms of heat transfer coefficient [\[26\]](#page-19-8). Non-uniform twisted inserts was equipped in conical horizontal tube to enhance heat transfer of S-CO₂ in $[27]$, the numerical results indicated that diverging conical tube with twisted inserts could enhance heat transfer by 43.1%. Obviously, the tube diameter, heat-flux mode, etc., have a significant effect on the heat transfer performance and flow characteristics of $S-CO₂$ in horizontal pipes. The heating and cooling conditions have opposite influences on the heat transfer performance of vertical S-CO₂ flows, and the heat transfer difference of S-CO₂ in horizontal flows between the conditions in which fluid was cooled and heated still needs to be further studied.

The existing literature indicates that the thermohydraulic properties of $S-CO₂$ flow are complex and its underlying mechanism of heat transfer is not fully recognised. In addition, previous studies have focused on analyses based on the first law of thermodynamics, and as Bejan [\[28\]](#page-19-10) pointed out, the second law (i.e., entropy production) would play a key role, since the amount of available energy lost during heat transfer is directly depended on the irreversibility. Heating conditions are very common in engineering applications, so in this work, we aim to comprehensively study the flow and heat transfer features of S-CO₂ horizontal flows under heating conditions based on the first- and second-law of thermodynamics, including exploring the effect of pipe diameter and heat flux, determining the laws of local irreversibility associated with heat flux, and revealing the heat transfer mechanism. The present work will deepen the field of understanding in the thermo-hydraulic characteristics of S-CO₂ flow and provide meaningful theoretical references for the design of heat exchangers and other related components using supercritical fluids as working fluids.

2 Model and Setting Conditions

2.1 Physical Model

[Fig. 1](#page-4-0) shows a 3D physical model of the horizontal tube used in this study, with a length of 1600 mm. To eliminate the influence of unsteady flow on the heat transfer performance, two sections with a length of 150 and 100 mm established in the outlet and inlet regions are assumed adiabatic, respectively. S-CO₂ flows up on the positive side of the *z*-axis, and gravity is set up on the negative side of the *y*-axis. To locate the position on the cross-section, the negative direction along the *y*-axis is set to 0°, and the positive direction along the *y*-axis is set to 180°. The pressure is set at 8.0 MPa, at which the corresponding T_{pc} is about 308 K. Consider the two diameters of 4 and 6 mm to assess their impact on heat transfer performance.

Figure 1: The physical model of the horizontal tube. The pipe length is set to 1600 mm, with two insulated areas of 150 and 100 mm at the inlet and outlet

2.2 Turbulence Model

It is assumed that the flow in the pipe is steady in the present work, ignoring the thermal resistance of the metal wall (much less than that of the fluid), as well as the heat transfer between the pipe wall and the environment (achieved through ideal insulation), the flow is three-dimensional and singlephase flow.

The governing equations are [\[29,](#page-19-11)[30\]](#page-19-12):

Equation of Continuity:

$$
\frac{\partial}{\partial x_j}(\rho u_j) = 0,\tag{1}
$$

Momentum conservation equation:

$$
\frac{\partial}{\partial x_j} \left(\rho u_i u_j \right) = \frac{\partial}{\partial x_j} \left[\mu_{\text{eff}} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{2}{3} \mu_{\text{eff}} \frac{\partial u_i}{\partial x_i} \right] - \frac{\partial P_i}{\partial x_i} + \rho g_i, \tag{2}
$$

Energy conservation equation:

$$
\frac{\partial}{\partial x_j} \left(\rho u_j c_p T \right) = \frac{\partial}{\partial x_j} \left(\lambda \frac{\partial T}{\partial x_j} \right) + \Phi, \tag{3}
$$

where *T* and *P* are temperature and pressure, *u* denotes velocity, *g* denotes the gravitational acceleration, c_p represents the specific heat at constant pressure, ρ denotes the density, λ denotes the thermal conductivity, μ_{eff} denotes the effective viscosity coefficient that can be obtained by summing the turbulent viscosity μ_1 and the molecular viscosity μ , Φ represents the energy dissipation related to viscosity.

The SST $k-\omega$ turbulent model is employed in this study, which is verified to be valid for S-CO₂ heat transfer and flow problems. This model has the advantage of *k-ω* turbulent model in accurate prediction in the vicinity of wall, and the robustness of *k*-*ε* model in the core region of the fluid (namely, the region that is far away from the pipe wall). The current simulations are conducted under steady state, so the specific dissipation rate *ω* and the turbulent kinetic energy *k* can be calculated through the following transport equations [\[30\]](#page-19-12):

Equation *for specific dissipation rate:*

$$
\frac{\partial}{\partial x_i} \left(\rho \omega u_i \right) + \frac{\partial}{\partial t} \left(\rho \omega \right) = \frac{\partial}{\partial x_j} \left(\Gamma_\omega \frac{\partial \omega}{\partial x_j} \right) + D_\omega + S_\omega + G_\omega - Y_\omega,
$$
\n(4)

Equation *for turbulent kinetic energy*:

$$
\frac{\partial}{\partial x_i} (\rho k u_i) + \frac{\partial}{\partial t} (\rho k) = \frac{\partial}{\partial x_j} \left(\Gamma_k \frac{\partial k}{\partial x_j} \right) + S_k + G_k - Y_k,
$$
\n(5)

where D_ω denotes the orthogonal divergence term, G_ω and G_k denote the production of ω and k , S_ω and S_k represent the source terms, Y_ω and Y_k represent the dissipations of ω and k related to turbulence, and Γ_{ω} and Γ_{k} represent the effective diffusivities of ω and k .

2.3 Grid and Model Validation

The mesh density of the established computational grid system in the boundary layer is enhanced to make sure that *y*⁺ *<* 1, otherwise the accuracy of simulation results cannot be guaranteed. Numerical calculations were conducted by using the software of ANSYS CFX, and the thermophysical properties of S-CO₂ (density, specific heat, and so on) were obtained from the National Institute of Standards and Technology (NIST) database [\[31\]](#page-19-13). To ensure high simulation accuracy, the high-resolution advection is employed. The universal convergence criterion is set to 1×10^{-6} , that is, when all the residuals of the variables in the governing equation above are less than this criterion, the calculation is automatically terminated and the result is considered to be convergent.

In this study, four mesh systems are used to test the grid independence. [Fig. 2](#page-5-0) shows the change in wall temperature (T_w) with the overall fluid temperature (T_b) for $P_{in} = 8.0$ MPa, $q = 50$ kW/m², $G = 400 \text{ kg/(m²·s)}$ and $T_{in} = 288.15 \text{ K}$. Taking the grid system with the largest number of elements (1.82 million) as the baseline case, one can see that when the mesh system whose element number is 1.24 million is employed, the average relative error of T_w from the baseline system is less than 0.1%, suggesting that when the element number is higher than 1.24 million, the increasing number of elements has little influence on the numerical results. Thereby, considering the computational resources and time as well as the acceptable simulation accuracy, the mesh system whose element number is 1.24 million is adopted in the present work.

Figure 2: Under the boundary conditions of $T_{in} = 288.15$ K, $P_{in} = 8.0$ MPa, $G = 400$ kg/(m²·s), and $q = 50 \text{ kW/m}^2$, the change of wall temperature in horizontal flow against the overall fluid temperature

Model verification is also carried out, and the results are obtained through experiments in [\[32\]](#page-19-14). When the conditions are set as follows: $T_{\text{in}} = 288.15 \text{ K}$, $P_{\text{in}} = 8.0 \text{ MPa}$, $G = 400 \text{ kg/(m²·s)}$, $q = 50 \text{ kW/m}^2$, and the numerical results obtained using the adopted model and empirical results are shown in [Fig. 3.](#page-6-0) It can be seen that the simulated value of T_w is almost consistent with the results in [\[32\]](#page-19-14), and the relative error is no more than 1.5% (on average). More information on the validation of the numerical model can be referred to our papers [\[33](#page-19-15)[–35\]](#page-19-16), which will not be repeated for simplicity. This shows that the model adopted in this work is reliable and its accuracy is acceptable.

Figure 3: Comparison of wall temperature between numerical results obtained using the adopted model in this study and experimental results reported in Reference $[32]$ with boundary conditions of T_{in} 288.15 K, $q = 50 \text{ kW/m}^2$, $P_{\text{in}} = 8.0 \text{ MPa}$, and $G = 400 \text{ kg/(m}^2 \cdot \text{s})$

2.4 Data Reduction

Full-wall heating conditions are considered in this work, with heat flux on the nonadiabatic (i.e., heated) section of the pipe whose *z*-value ranging from 150 to 1500 mm as shown in [Fig. 1.](#page-4-0) The inlet pressure and inlet temperature are set to 8.0 MPa and 288.15 K, respectively. Various slices are established in the heated region (between $z = 150$ mm and $z = 1500$ mm) along the positive direction of the *z*-axis (flow direction), with an interval space of 25 mm. Local parameters, including entropy generation, fluid bulk temperature and wall temperature, can be obtained based on these transverse slices (sections). Accordingly, the local heat transfer coefficient (h_h) can be obtained:

$$
h_{\rm b} = \frac{q}{T_{\rm w} - T_{\rm b}},\tag{6}
$$

where *q* denotes the local heat flux, and subscripts "w" and "b" denote pipe wall and fluid bulk, respectively. Local wall temperature T_w is the linear average temperature based on the intersection line of the inner wall and the transverse section, and the fluid bulk temperature T_b on the transverse section can be written as:

$$
T_{\rm b} = \frac{\int_A \rho u c_{\rm p} T \mathrm{d}A}{\int_A \rho u c_{\rm p} \mathrm{d}A}.\tag{7}
$$

Bejan [\[28\]](#page-19-10) stated that the local irreversibility in heat transfer processes is associated with two factors, i.e., heat transfer (temperature difference) and flow resistance, these are written as:

$$
S_{g,\Delta T} = \left(\left(\frac{\partial T}{\partial x} \right)^2 + \left(\frac{\partial T}{\partial y} \right)^2 + \left(\frac{\partial T}{\partial z} \right)^2 \right) \frac{\lambda}{T^2},\tag{8}
$$

$$
S_{g,\Delta P} = \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}\right) \frac{\partial u_i}{\partial x_j} \frac{\mu}{T}.
$$
\n(9)

Under the action of gravity, the drastic changes in thermophysical properties lead to the buoyancy effect, which will have an important impact on the heat transfer performance and flow characteristics of S-CO2. To quantitatively assess the buoyancy effect, a dimensionless criterion is usually employed for $S-CO₂$ in horizontal flows, and the criterion is written as [\[36,](#page-20-0)[37\]](#page-20-1):

$$
Gr = \frac{(\rho_{\rm w} - \rho_{\rm b})\,\rho_{\rm b}gd^3}{\mu_{\rm b}^2},\tag{10}
$$

$$
Ri = \frac{Gr}{Re_b^{2.0}} = \frac{(\rho_w - \rho_b) \rho_b g d}{G^2},
$$
\n(11)

here *d* denotes the pipe's diameter, *Re* denotes the Reynolds number of flows, *Gr* denotes the Grashof number. It is generally acknowledged that the effect of buoyancy needs to be considered in the case of *Ri* > 0.001, and the effect becomes more important with the increase of *Ri*.

The drastic change of thermophysical properties may also lead to the effect of thermal acceleration, which impacts the heat transfer features of $S-CO₂$ greatly by influencing the boundary layers. The criterion of Kv is employed to assess the influence of thermal acceleration for $S-CO₂$ flow in this study [\[38\]](#page-20-2):

$$
Kv = \frac{4q d\alpha}{Re_b^2 \mu_b c_p},\tag{12}
$$

where α is the coefficient of thermal expansion. In general, when the criterion of Kv is less than 3 \times 10^{-6} , the thermal acceleration effect can be neglected in the studies of S-CO₂ flows.

3 Results and Discussion

Notable differences can be observed in flow and heat transfer features of S-CO₂ flows when the boundary conditions are slightly different, since the thermophysical properties vary drastically. In this section, the influences of pipe diameter and heat-flux conditions on the heat transfer features are investigated, and the irreversibilities associated with heat transfer and pressure drop in the S-CO₂ flows are explored, along with a detailed discussion on the underlying mechanism.

3.1 Influence of Pipe Diameter

To analyse the effect of the tube diameter on the thermal-hydraulic features in S-CO₂ flows, two tubes of 4 and 6 mm are considered, and the other geometric parameters of the pipe are kept the same as shown in [Fig. 1.](#page-4-0) To keep the same inlet Reynolds number for the two pipes, the inlet mass flux is set to be 400 kg/(m²·s) for the 4 mm pipe, and 300 kg/(m²·s) for the 6 mm pipe, respectively. The other conditions are set as follows: $T_{in} = 288.15 \text{ K}$, $q = 50 \text{ kW/m}^2$, and $P_{in} = 8.0 \text{ MPa}$. Variations of T_{w} and h_b in response to T_b along the flow direction (in the positive direction of the *z*-axis in [Fig. 1\)](#page-4-0) are presented in [Fig. 4.](#page-8-0) In can be seen from [Fig. 4a](#page-8-0) that the T_w gradually increases upstream of where T_b is ∼1 K below T_{pc} and then increases significantly. Moreover, T_{w} is ∼24 K higher in the 6 mm pipe than the 4 mm pipe on average. [Fig. 4b](#page-8-0) shows that the h_b generally increases after a slight decrease in the entrance of the diabatic section, and reaches a peak where T_b is ∼1 K below T_{pc} . The heat transfer coefficient (h_b) is increased by an average of 1.3 times in the 4 mm pip, and the peak value is about 3 times as much as that in the 6 mm pipe.

Figure 4: In two pipes with $d = 4$ mm and 6 mm, (a) wall temperature and (b) heat transfer coefficient against the bulk fluid temperature in the flow direction (namely, the positive direction of the z-axis in [Fig. 1\)](#page-4-0). The mass flux is set to 300 and 400 kg/(m^2 -s) to maintain the same inlet Reynolds number

The variation of entropy generation caused by heat transfer $(S_{g,\Delta T})$ and flow resistance $(S_{g,\Delta P})$ with the fluid bulk temperature in the two pipes is demonstrated in [Fig. 5.](#page-8-1) In the 6 mm pipe, $S_{\gamma\Lambda T}$ increases significantly to a peak where T_b is ∼292 K, tends to be stable after a slight decrease, and drops drastically downstream of $T_b = 303$ K. In the 4 mm pipe, $S_{g, \Delta T}$ increases greatly to a peak where T_b is ∼294 K, decreases to a trough when T_b is ∼1 K below T_{pc} , and increases significantly downstream of where *T*_b is ∼1 K lower than *T*_{pc}, and *S*_{g,∆*T*} is ∼67% lower on average in the 4 mm pipe relative to the 6 mm pipe. [Fig. 5b](#page-8-1) shows that the two flows deliver similar changing trends in $S_{g,\Delta P}$, which increases slightly and then significantly downstream of where T_b is ∼2 K below T_{pc} , and $S_{g, \Delta P}$ is ∼1.1 times higher on average in the 4 mm pipe. It can be seen that $S_{g,\Delta T}$ is higher in 6 mm pipe upstream of the pseudo-critical point, while $S_{g, \Delta P}$ is higher in the 4 mm pipe over the whole diabatic region, and also, $S_{g, \Delta T}$ is more than 100 times higher than $S_{g, \Delta P}$ in both flows, indicating that heat transfer is the dominant factor for local entropy generation.

Figure 5: Variation of entropy generations that caused by (a) temperature difference, and (b) flow resistance, with bulk fluid temperature in the positive *z*-axis in [Fig. 1](#page-4-0) in the two pipes with $d = 4$ mm and 6 mm. Mass fluxes are set to be 300 and 400 kg/(m²·s), respectively, to keep the same inlet Reynolds number, and the other conditions are: $T_{in} = 288.15 \text{ K}$, $P_{in} = 8.0 \text{ MPa}$, and $q = 50 \text{ kW/m}^2$

[Fig. 6](#page-9-0) shows the relationships of criteria of thermal acceleration and buoyancy effect with T_b in the two pipes. Generally, *Ri* decreases in both flows as T_b rises, and the value of *Ri* is larger than 0.001 (the threshold value) in the whole diabatic region. Of note is that *Ri* is higher in the 6 mm pipe than the 4 mm pipe, so the buoyancy effect is more remarkable in the larger pipe. [Fig. 6b](#page-9-0) shows that *Kv* increases firstly and then decreases as T_b rises, there exists a peak in the region at which T_b is about 2 K lower than T_{pc} , it is worth noting that *Kv* is always lower than the threshold value of 3×10^{-6} , indicating that the effect of thermal acceleration can be ignored.

Figure 6: The relations of criteria of (a) Buoyancy effect and (b) thermal acceleration with the fluid bulk temperature in the two tubes of $d = 4$ mm and 6 mm. The mass flux is set to 300 and 400 kg/(m²·s) to maintain the same inlet Reynolds number

3.2 Influence of Imposed Heat Flux

Three types of heat fluxes of 50, 75 and 100 kW/m², are considered to investigate the influence on flow and heat transfer features of horizontal S-CO₂ flows. The pipe with a diameter of 4 mm is selected, with other conditions fixed at $T_{\text{in}} = 288.15 \text{ K}$, $P_{\text{in}} = 8.0 \text{ MPa}$ and $G = 400 \text{ kg/(m²·s)}$.

[Fig. 7](#page-10-0) shows that T_w increases gradually with T_b upstream of T_{pc} , and rises greatly downstream of T_{pc} when the heat fluxes are 50 and 75 kW/m². While under the highest heat flux of $q = 100 \text{ kW/m}^2$, *T*_w rises to a peak at which T_b is ∼300 K and decreases to a trough where T_b is ∼2 K above T_{pc} , and increases again along the flow direction. [Fig. 7b](#page-10-0) shows that under the lowest heat flux of 50 kW/m2 , *h*^b increases after a slight decrease in the entrance of the diabatic section, and reaches a peak where T_b is 1.5 K below T_{pc} , and drops sharply downstream of T_{pc} . When the heat fluxes are 75 and 100 kW/m², h_b decreases to a trough and increases to a peak when T_b approaches T_{pc} , and also decreases downstream of T_{pc} . Of note is that the trough and the peak of h_b shift to the positions with a higher T_b as heat flux rises. More precisely, the peak of h_b occurs where T_b is 1.5 K below T_{pc} with *q* $= 50 \text{ kW/m}^2$, where T_b is 0.5 K below T_p with $q = 75 \text{ kW/m}^2$, and where T_b is ∼2 K above T_p with $q = 75 \text{ kW/m}^2$. $= 100 \text{ kW/m}^2$, respectively. The peak of h_b rises by 1.4 times with the reduction of heat flux from 100 to 50 kW/m2 .

Figure 7: (a) Wall temperature and (b) heat transfer coefficient against bulk fluid temperature in positive *z*-axis in the 4 mm pipe with heat fluxes of 50, 75 and 100 kW/m2

The variation of entropy generation caused by temperature difference $(S_{g, \Delta T})$ and flow resistance $(S_{g,\Delta P})$ with the fluid bulk temperature is presented in [Fig. 8.](#page-10-1) [Fig. 8a](#page-10-1) shows that $S_{g,\Delta T}$ generally reduces as T_b grows with heat fluxes of 75 and 100 kW/m². When the heat flux is 50 kW/m², $S_{g,\Delta T}$ increases and decreases afterwards, and then increases again with the increase in T_b . Of note is that $S_{g,\Delta T}$ at $q =$ 100 kW/m² is more than 4 times higher than that at $q = 50$ kW/m² on average. [Fig. 8b](#page-10-1) shows that $S_{g, \Delta P}$ increases steadily upstream of T_{pc} and increases significantly downstream of T_{pc} , and almost coincides with the three heat fluxes, indicating the effect of heat flux on $S_{g, \Delta P}$ is very little. The heat flux has a more significant effect on $S_{g, \Delta T}$, and also, $S_{g, \Delta T}$ is more than 3 orders of magnitude higher than $S_{g, \Delta P}$ under the conditions selected in the present work.

Figure 8: Variations of entropy generations caused by temperature difference and flow resistance against the fluid bulk temperature in positive *z*-axis in [Fig. 1](#page-4-0) with heat fluxes of 50, 75 and 100 kW/m2

From [Fig. 9a,](#page-11-0) one can see that the buoyancy effect criterion *Ri* drops significantly upstream of where T_b is ∼2 K above T_{pc} , and decreases to nearly zero downstream of T_{pc} . Moreover, *Ri* is always above the threshold value (0.001) in the region at which T_b is less than 340 K, and it becomes higher as heat flux rises, this demonstrates that the buoyancy effect must be considered and is strengthened as the heat flux increases. [Fig. 9b](#page-11-0) shows that the thermal acceleration criterion *Kv* yields a similar

changing trend for the three heat fluxes, which increases firstly and then decreases, there exists a peak in the region at which T_b is ∼2 K below T_{pc} , with a drastic drop observed near the pseudo-critical point. One can see that the criterion *Kv* is always lower than 3×10^{-6} (the threshold value), indicating that the effect of thermal acceleration can be ignored under the conditions selected in the present work, although the influence of heat flux on the *Kv* value is significant.

Figure 9: The relationships of change of (a) Buoyancy effect criterion, and (b) thermal acceleration criterion against bulk temperature in positive *z*-axis in [Fig. 1,](#page-4-0) the three heat fluxes are 50, 75 and 100 kW/m2

3.3 The Difference between Heat Transfer and the Underlying Mechanism

The results above show that the criterion of the buoyancy effect is much larger than the critical value (which is 0.001), indicating that the buoyancy effect will greatly affect the local thermalhydraulic features of $S-CO₂$ flows. To further identify the difference of local heat transfer performance in horizontal S-CO₂ flows at heating conditions, average h_b based on full wall, top-half wall and bottom-half wall as a function of T_b is presented in [Fig. 10](#page-11-1) under conditions of $T_{in} = 288.15$ K, $G = 400 \text{ kg/(m²·s)}$, $q = 50 \text{ kW/m²}$, and $P_{\text{in}} = 8.0 \text{ MPa}$.

Figure 10: Average heat transfer coefficient based on full, top-half, and bottom-half walls against fluid bulk temperature in positive *z*-axis in [Fig. 1](#page-4-0) at conditions of $T_{in} = 288.15$ K, $G = 400$ kg/(m²·s), $q =$ 50 kW/m², and $P_{\text{in}} = 8.0 \text{ MPa}$

There exists a peak of where T_b is ∼1 K below T_{pc} under the three conditions, and h_b based on the bottom-half wall of the pipe is far higher than that based on the top-half wall (more than 2 times in peak value) and full wall. This indicates that there exists a huge difference in the performance of heat transfer on the same transverse section, and the heat transfer is augmented in the vicinity of the lower wall of pipe.

To explore the underlying mechanism in the heat transfer difference shown in [Fig. 10,](#page-11-1) four transverse sections whose *z* values are 800, 900, 1000 and 1100 mm are selected to analyse the distributions of key parameters including the fluid temperature and velocity in the flow.

The temperature contours on the four transverse sections are shown in [Fig. 11,](#page-12-0) indicating that temperature distributions are severely nonuniform and the fluid temperature in the vicinity of the upper wall of pipe is higher than that at the bottom. Moreover, the high-temperature region at the top enlarges along the flow direction, and T_w is higher in the top region than in the bottom region.

Figure 11: Temperature contours on four transverse sections whose *z* values are (a) 800 mm, (b) 900 mm, (c) 1000 mm, and (d) 1100 mm in the 4 mm pipe, at conditions of $T_{\text{in}} = 288.15$ K, $G =$ 400 kg/(m²·s) , $q = 50 \text{ kW/m²}$, and $P_{\text{in}} = 8.0 \text{ MPa}$

[Fig. 12](#page-13-0) shows the velocity vectors on the considered transverse sections, which indicate that secondary flows (represented by two vortexes) are formed due to the buoyancy effect. A little change of temperature would lead to a large variation of density on the cross-section due to the drastic variations of thermophysical properties of $S-CO_2$, and the variable density on the cross-section leads to buoyancy

effect under the influence of gravity. The secondary flow as well as the jet flow are formed in the process that the lighter fluid moves upwards and the heavier fluid moves downwards, which affects heat and flow characteristics greatly. The fluid flows away from the region near the top wall and moves towards the region near the bottom wall, generating a jet flow in the vicinity of the lower wall of pipe, and the velocity is higher at the bottom than that at the top. Moreover, the velocity increases (from 0.73 m/s at $z = 800$ mm to 0.99 m/s at $z = 1100$ mm) in the positive direction of *z*-axis due to the density decrease at heating conditions.

Figure 12: Velocity vectors on four transverse sections whose *z* values are (a) 800 mm, (b) 900 mm, (c) 1000 mm, and (d) 1100 mm in the 4 mm pipe, at conditions of $T_{in} = 288.15$ K, $G = 400$ kg/(m²·s), $q =$ 50 kW/m², and $P_{\text{in}} = 8.0 \text{ MPa}$

To further explore the relationship between the two fields of velocity and temperature gradient, the field synergy principle that proposed by Guo et al. [\[39,](#page-20-3)[40\]](#page-20-4) is adopted, in this principle the dimensionless Nusselt number can be written as:

$$
Nu = RePr \int_0^1 (\overline{\nabla} \mathbf{T} \cdot \overline{\mathbf{U}}) d\overline{y},
$$
\n(13)

$$
\overline{\mathbf{U}} \cdot \overline{\nabla \mathbf{T}} = |\overline{\mathbf{U}}| \cdot |\overline{\nabla \mathbf{T}}| \cdot \cos(\beta),\tag{14}
$$

with:

$$
0^{\circ} \le \beta \le 180^{\circ},\tag{15}
$$

where $\overline{\nabla}$ **T** represents the field of the temperature gradient, \overline{U} represents the velocity field, *Pr* denotes the Prandtl number, and β represents the synergy (coordination) angle between $\overline{\nabla T}$ and \overline{U} . For the convenience of compassion, the synergy angle β is converted into the range of 0° to 90°, so a smaller β means that $\overline{\nabla T}$ and $\overline{\nabla}$ are more synergetic, and the heat transfer performance is better.

The synergy angle contours on the four transverse sections whose *z* values are 800, 900, 1000 and 1100 mm are presented in [Fig. 13.](#page-14-0) This indicates that β in the near-wall region at the bottom is smaller while is higher at the top. The fluid flow in the region below the two vortices is the most intense, forming a jet flow towards the bottom wall. Thus, the temperature gradient and velocity fields in this region are also the most synergetic, resulting in the smallest *β*. On the contrary, the near-wall region at the top is less affected by the vortices, as the two circulation flows collide, and the temperature gradient and velocity fields are the least synergetic, resulting in the largest β . Therefore, the heat transfer is significantly improved in the near-wall region at the bottom of the pipe.

Figure 13: Synergy angle between fields of velocity and temperature gradient on four transverse sections whose *z* value is (a) 800 mm, (b) 900 mm, (c) 1000 mm, and (d) 1100 mm in the 4 mm pipe, at conditions of $T_{\text{in}} = 288.15 \text{ K}$, $G = 400 \text{ kg/(m}^2 \cdot \text{s)}$, $q = 50 \text{ kW/m}^2$, and $P_{\text{in}} = 8.0 \text{ MPa}$

The contours of entropy generations caused by temperature difference and flow resistance at the cross-section of $z = 1000$ mm are demonstrated in [Fig. 14,](#page-15-0) showing that large $S_{g, \Delta T}$ and $S_{g, \Delta P}$ are mainly identified in the near-wall region. It can be seen that $S_{g, \Delta T}$ is distributed nonuniformly, and large $S_{g,\Delta T}$ is mainly concentrated in the vicinity of the upper wall of pipe, while it is less distributed at the bottom. By comparing [Figs. 10](#page-11-1) and [14,](#page-15-0) it can be concluded that h_b and $S_{g,\Delta T}$ have opposite distribution characteristics, namely, the region with higher h_b yields less irreversible loss caused by heat transfer. It also can be seen that $S_{g, \Delta P}$ is evenly distributed in the near-wall region, indicating that it is less affected by the buoyancy effect. In addition, [Fig. 14](#page-15-0) demonstrates that the irreversible losses in the core region are much lower than that in the near-wall region.

Figure 14: Entropy generations caused by (a) temperature difference, and (b) flow resistance at the cross-section at location of $z = 1000$ mm in the 4 mm pipe, at conditions of $T_{in} = 288.15$ K, $G =$ 400 kg/(m²·s) , $q = 50 \text{ kW/m²}$, and $P_{\text{in}} = 8.0 \text{ MPa}$

[Fig. 15](#page-16-0) shows the distribution of radial density, specific heat capacity, thermal conductivity, and turbulent kinetic energy on a cross-section of $z = 1100$ mm. [Fig. 15a](#page-16-0) shows that the density generally decreases from the bottom of the pipe to the top due to the concentration of low-density fluid in the vicinity of the upper wall, while high-density fluid is concentrated in the vicinity of the lower wall due to the effect of buoyancy. The specific heat decreases and then rises from core to wall to peak, dropping substantially in the near-wall region along the $\theta = 0^{\circ}$ direction and $\theta = 45^{\circ}$ direction, as shown in [Fig. 15b,](#page-16-0) while remaining almost constant in the region where is far away from the wall (i.e., the core region), dropping in the direction from fluid core to the wall the at $\theta = 90^{\circ}$, $\theta = 135^{\circ}$, as well as $\theta = 180^{\circ}$. In general, in the near-wall region, the specific heat capacity decreases from the bottom to the top, indicating that the fluid has an enhanced heat transfer capacity in the vicinity of the lower wall. [Fig. 15c](#page-16-0) shows that thermal conductivity generally decays from bottom to top, with thermal conductivity from core to wall remaining almost constant at $\theta = 0^{\circ}$, $\theta = 45^{\circ}$, as well as θ $= 90^{\circ}$, while thermal conductivity from core to wall gradually decreases at $\theta = 135^{\circ}$ and $\theta = 180^{\circ}$. [Fig. 15d](#page-16-0) shows that in the core of fluid, the turbulent energy generally decreases from top to bottom, while in the vicinity of wall, the turbulent energy decreases from bottom to top. It can be seen that the distribution of local parameters is seriously uneven, and the density, specific heat, thermal conductivity and turbulent kinetic energy in the vicinity of wall from the upper wall to the lower wall, indicating that the heat transfer in the vicinity of lower wall is enhanced, and the heat transfer in the vicinity of the upper wall deteriorates.

Figure 15: Radial distributions of: (a) density, (b) specific heat, (c) thermal conductivity, and (d) turbulent kinetic energy along directions of $\theta = 0^{\circ}$, $\theta = 45^{\circ}$, $\theta = 90^{\circ}$, $\theta = 135^{\circ}$ and $\theta = 180^{\circ}$ at the cross-section of $z = 1100$ in the 4 mm pipe, at $G = 400$ kg/(m²·s), $T_{in} = 288.15$ K, $P_{in} = 8.0$ MPa, and $q = 50 \text{ kW/m}^2$

4 Conclusions

The flow and heat transfer features of a heated horizontal $S-CO₂$ flow are numerically investigated, taking into account the flow of a pressure of 8.0 MPa through two pipes of different diameters (4 and 6 mm), with two mass fluxes of 300 and 400 kg/(m^2 ·s), and with three heat fluxes of 50, 75 and 100 kW/m2 . The effects of pipe diameter and heat flux density on thermal-hydraulic performance and irreversible loss were investigated, and the underlying mechanism was analysed.

- (1) The peak heat transfer coefficient of 4 mm pipeline is about 3 times higher than that of 6 mm pipeline, while the average irreversible heat transfer loss of 4 mm pipeline is about 67% lower, and the average irreversible loss of flow resistance is 1.1 times higher.
- (2) As the heat flux decreases from 100 to 50 kW/m², the peak heat transfer coefficient in the 4 mm pipe increases by a factor of 1.4, while the frictional irreversibility is almost unaffected.

When the wall heat flux is 100 kW/m^2 , the irreversible loss due to heat transfer is more than four times higher than when the heat flux of 50 kW/m² is adopted.

- (3) In this study, the thermal acceleration effect is negligible, while the buoyancy effect is important because it creates a secondary flow that significantly affects the flow and heat transfer features in the pipe.
- (4) The distribution of local parameters is seriously uneven, and the peak of heat transfer coefficient in the vicinity of the lower wall of pipeline is more than 2 times higher than that at the top. Two vortices on the cross-section (secondary flow) are formed under buoyancy, and the jet is observed in the vicinity of the lower wall, which improves the coordination degree between the two fields of velocity and temperature gradient in this region, thereby enhancing heat transfer.
- (5) Two circulating flows collide in the vicinity of the upper wall, weakening the synergy of the two fields of the velocity and temperature gradient, leading to a deterioration of heat transfer in that region. The increase of heat transfer performance corresponds to the decrease of irreversible loss related to temperature difference, and the irreversible loss caused by flow resistance is less affected by buoyancy effect.
- (6) The distribution of local parameters is also seriously uneven, from the bottom to the top of the pipe, the specific heat capacity, thermal conductivity and turbulent kinetic energy in the near-wall region are reduced, demonstrating that heat transfer is intensified in the vicinity of the lower wall.
- (7) This study contributes to a better understanding of the heat transfer performance and flow characteristics of heating S-CO₂ flow in horizontal pipelines, and further enriches and develops the heat transfer theory of supercritical fluids by describing singular heat transfer phenomena and revealing related mechanisms. This provides meaningful reference and theoretical guidelines for the design and optimization of heat exchange equipment and other related components using supercritical fluids as working fluids.

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