

DOI: 10.32604/cmes.2023.025501

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Reliability-Based Topology Optimization of Fail-Safe Structures Using Moving Morphable Bars

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ABSTRACT

This paper proposes an effective reliability design optimization method for fail-safe topology optimization (FSTO) considering uncertainty based on the moving morphable bars method to establish the ideal balance between cost and robustness, reliability and structural safety. To this end, a performance measure approach (PMA)-based double-loop optimization algorithm is developed to minimize the relative volume percentage while achieving the reliability criterion. To ensure the compliance value of the worst failure case can better approximate the quantified design requirement, a p-norm constraint approach with correction parameter is introduced. Finally, the significance of accounting for uncertainty in the fail-safe design is illustrated by contrasting the findings of the proposed reliability-based topology optimization (RBTO) method with those of the deterministic design method in three typical examples. Monte Carlo simulation shows that the relative error of the reliability index of the optimized structure does not exceed 3%.

KEYWORDS

Topology optimization; fail-safe design; uncertainty; reliability-based topology optimization; moving morphable bars

1 Introduction

Topology optimization (TO) can generate innovative designs without any assumptions about structural connectivity, which makes it a potent and helpful drive for reducing structural weight while satisfying specific performance requirements. Nowadays, TO has been successfully implemented in numerous fields and the available reviews on the most recent developments can be found in literature reviews [1,2] and references therein.

Due to the absence of structural redundancy and disregard for the influence of uncertain parameters, the resulting resembled statically determinate structures obtained by conventional TO are susceptible to local stiffness loss. FSTO aims to design structures that can operate in a damaged



state. Fail-safe structures can overcome the disadvantage that traditional structures are easily affected by local stiffness loss due to the absence of structural redundancy. Jansen et al. [3] introduced the failsafe idea into TO for the first time. They demonstrated the influence of local failure on the typical design problem of cantilever beam and compliant mechanism, which may lead to the complete loss of displacement inversion capability of compliant mechanism. Therefore they suggested a streamlined model of local damage and a robust worst-case approach for designing fail-safe structures, in which the complex process of damage is approximated by removing material stiffness in fixed square patches with a prescribed size. In addition, they presented other ways to reduce the computational time of the proposed approach, such as sequential formulation, perturbation techniques, reanalysis techniques, and active set strategies. Zhou et al. [4] generalized the work of FSTO to establish a rigorous mathematical design framework and computationally feasible solution for 3D structures. Stolpe [5] extended truss TO to accommodate fail-safe requirements, where the damage scenarios are modelled by removing a set of bars' material stiffness or by degrading the member regions. Lüdeker et al. [6] presented a stress-constrained formulation for fail-safe optimization using beam elements with varying thicknesses, where the damage scenario is modelled by removing one beam element at a time. Wang et al. [7] incorporated the von Mises stress into FSTO, where damage situations are simulated by removing the material stiffness of certain patches whose von Mises stress exceeded the permissible value. Du et al. [8] integrated fail-safe requirements into the field of frequency TO via the independent continuous mapping (ICM) method, which prevents the fundamental frequency of optimized fail-safe structures from being vulnerable to local failure. Du et al. [9] presented FSTO model for minimizing the weight of continuum structures with stress and displacement constraints based on the ICM method and the dual sequence quadratic programming (DSQP) algorithm. Kranz et al. [10] developed a volume-restricted stress-based FSTO formulation for minimizing the worst-case von Mises stress. Peng et al. [11] developed a volume-minimization fail-safe optimization formulation with displacement restrictions by adopting the reciprocal type variables. To circumvent the finite element (FE) mesh dependency in existing methods for FSTO, Hederberg et al. [12] employed socalled moving morphable components (MMCs) [13,14] to simulate damage while using the variable density method to implement TO. Smith et al. [15] employed a geometry projection method for FSTO, in which local failure is mimicked by removing each geometric component. Dou et al. [16] presented some notable findings of extending stress-constrained truss optimization to accommodate fail-safe requirements through a three-bar example. Then they incorporated two additional failure modes of thickness degradation and local damage into the FSTO of 2D frame structures subject to stress and eigenfrequency criteria [17]. As an alternative to defining a damage formulation for boosting the redundancy and damage tolerance, Wu et al. [18] and Dou [19] integrated fail-safe criterion into TO by applying local volume constraints.

It is observed that the aforementioned research on FSTO is conducted under deterministic conditions, i.e., deterministic topology optimization (DTO), which means it ignores the effect of inherent uncertainty factors, such as material parameters, on the design optimization of fail-safe structures. However, uncertainties of material properties and load are essential and critical features in FSTO structures, particularly in large structures and lifeline engineering structures. In addition, as highlighted in [4,6,20,21], fundamental engineering structures have high requirements for robustness, FSTO reliability, and structural safety. The underlying concepts behind FSTO and reliability-based or robust TO are very different and should not be confused, even though these approaches may provide comparable results. In FSTO, the location of failure cases or degradation was known red, and the corresponding areas were removed entirely. Robust and reliability-based TO are two strategies for

coping with unknown parameters involved in structural optimization. Robust TO aims to enhance the capacity of structures to withstand changes in uncertain parameters [22,23]. RBTO entails reliability analysis and ensures the fulfillment of the design conditions with the intended target reliability level when some parameters involved in structural responses are of random nature [24,25]. Reliability design has been extended to many fields of topology optimization, including geometrically nonlinear [26,27], dynamic [28,29], stress-constrained [30], thermal-mechanical coupling [31], composite structures [32], etc. The approximate reliability methods based on Taylor series expansion, including the first-and second-order reliability methods (FORM and SORM), have proved to be accurate methods for estimating the failure probability in RBTO [33-35]. It seems reasonable to integrate these concepts to mitigate potential performance degradation during service and provide more robust and reliable fail-safe designs. Currently, some researchers have made some efforts in this regard. Long et al. [36] introduced a resilient TO formulation for continuum structures considering both the unpredictability of the damage site and the uncertainty of loading direction and amplitude. A technique of a robust TO of fail-safe systems that considers the occurrence and size of failure as sources of uncertainty has been proposed by Martínez-Frutos et al. [37]. In contrast to using a worst-case approach, Martínez-Frutos et al. [20] also proposed a risk-averse level-set method for TO of fail-safe structures, in which the probability of a failure scenario occurring is coupled with the formulation using a probabilistic framework. Cid et al. [38] introduced a novel probability-damage technique based on available information regarding the chance associated with individual partial collapses.

It can be seen that most of the existing literature on fail-safe design is conducted based on the implicit density method and level-set method. The implicit geometric/topological representation is not compatible with modern computer-aided design (CAD) modelling systems. In this paper, we propose an explicit reliability-based TO methodology for handling the design problem of fail-safe structures under uncertainty using explicit moving morphable bars (MMB), aiming to achieve an optimal balance between cost, robustness, reliability, and structural safety. This explicit method supports explicit geometric information due to the use of geometric parameters as design variables. With the help of this geometric information, we can easily reconstruct the optimized structure in a CAD system and then generate file formats that can be read by CNC (computer numerical control) machining or additive manufacturing, such as STL files. Another notable benefit of using MMB method is that it can facilitate the implementation of minimum constraint control and improve the manufacturability of optimized structures. Besides, using the MMB method to implement topology optimization involves relatively fewer design variables, which can improve the calculation efficiency of optimization iteration. The rest of this paper is organized as follows. Section 2 details the deterministic FSTO based on moving morphable bars. Section 3 formulates RBTO of fail-safe structures considering uncertainty, followed by the design sensitivity analysis in Section 4. The suggested RBTO method for fail-safe structures is demonstrated in Section 5 through a series of numerical tests. Section 6 comes to a conclusion.

2 DTO of Fail-Safe Structures Based on Moving Morphable Bars

2.1 Density Distribution in Moving Morphable Bars

As opposed to the traditional pixel or node point-based implicit TO approach, Here we use the MMB method [39–41] to conduct TO due to its convenience in implementing minimum size control. The MMB method uses a set of discrete geometric bars as the fundamental building blocks. The geometrical parameters describing the position and size of these MMBs are taken as the design variables. As shown in Fig. 1, each discrete bar is determined by 5 parameters: coordinates of the centers k1, k2 of the two semicircles at the ends of the bar (x_{k1}, y_{k1}) and (x_{k2}, y_{k2}) , and the bar thickness

 t_k ($t_k = 2r_k$; r_k is the radius of the semicircular ends). Inspired by the seminal works [42,43], to escape the hassle of remeshing due to bar movement, these geometric bars are then projected onto a density field via a differentiable function in Eq. (1). The element density ρ_e of element *e* is thus defined as [39,40]

$$\rho_e = 1 - \prod_{k=1}^{N_b} \frac{1}{1 + \exp\left[-\beta(d_{ek} - r_k)\right]} \tag{1}$$

where N_b is the number of MMBs. β refers to the rate parameter, which determines how close ρ_e is to 0 or 1. d_{ek} is defined as the euclidean distance between the center of the *e*th element and the *k*th bar, given by

$$d_{ek} = \begin{cases} \sqrt{\left(\overline{x}_{k1} - \overline{x}_{e}\right)^{2} + \left(\overline{y}_{k1} - \overline{y}_{e}\right)^{2}} & \text{if } \overline{x}_{e} < \overline{x}_{k1} \\ \frac{||\mathbf{A}_{ek}||}{l_{k}} & \text{if } \overline{x}_{k1} \le \overline{x}_{e} \le \overline{x}_{k2} \\ \sqrt{\left(\overline{x}_{k2} - \overline{x}_{e}\right)^{2} + \left(\overline{y}_{k2} - \overline{y}_{e}\right)^{2}} & \text{if } \overline{x}_{e} > \overline{x}_{k2} \end{cases}$$

$$(2)$$



Figure 1: Geometry of a moving bar

In Eq. (2), l_k and \mathbf{A}_{ek} are defined as

$$l_{k} = \sqrt{(\bar{x}_{k2} - \bar{x}_{k1})^{2} + (\bar{y}_{k2} - \bar{y}_{k1})^{2}}$$
(3)

$$\mathbf{A}_{ek} = \begin{bmatrix} \overline{x}_{k1} - \overline{x}_{e} & \overline{y}_{k1} - \overline{y}_{e} \\ \overline{x}_{k2} - \overline{x}_{e} & \overline{y}_{k2} - \overline{y}_{e} \end{bmatrix}$$
(4)

Note that $(\overline{x}, \overline{y})$ represents local coordinates, where the x-axis is defined as parallel to the axis of the MMBs. x_e and y_e denote the center coordinates of the *e*th element.

For clarity, in Fig. 2, the figure on the left shows the pattern of "I \heartsuit U" made up of geometric bars, and the figure on the right shows the density distributions of "I \heartsuit U" mapped from Eq. (1) with $\beta = 8$. From Eq. (1) and Fig. 2, we can conclude that the value of ρ_e is one if the center of the *e*th element is within one of MMBs. Otherwise, ρ_e is zero.



Figure 2: The pattern of "I ♡ U" made up of geometric bars, and its density distributions

2.2 DTO of Fail-Safe Structures

Here we adopted the simplified local damage model reported in [3] to implement FSTO. An adequate set of *m* patch removal scenarios are used to depict local failure. As illustrated in Fig. 3 for a quadrate failure zone containing a patch of elements $P^{(i)}$ where local failure caused by an accidental event happens. Then the material interpolation scheme for designing fail-safe structures can be defined as

$$E_{e}^{(i)} = E_{\min} + \left(\overline{\rho}_{e}^{(i)}\right)^{p} \left(E_{0} - E_{\min}\right)$$
(5)

where $\rho_{e}^{(i)}$ is defined as

$$\overline{\rho}_{e}^{(i)} = \begin{cases} \rho_{e} & \text{if } e \in N \setminus P^{(i)} \\ \rho_{\min} & \text{if } e \in P^{(i)} \end{cases}$$
(6)

where E_0 represents Young's modulus of solid material. N denotes the elements index set. p is the penalization factor, typically chosen p = 3. ρ_{\min} is a small term used to prevent the stiffness matrix from becoming singular.



Figure 3: Removal of a quadrate patch of material

As with the pioneering works integrating local failure into continuum TO [3,4], we use compliance as the performance measure of fail-safe structures involving failure scenarios. However, unlike performing the volume-constrained compliance minimization in [3,4], we treat the compliance corresponding to the worst failure case as the design constraint while minimizing the volume fraction of the structure. This means that we can achieve the purpose of weight reduction while meeting the quantified design requirements. Consequently, the DTO model for fail-safe structures is formulated as

find :
$$\mathbf{z} = \{\mathbf{z}_{1}, \mathbf{z}_{2}, \cdots, \mathbf{z}_{N_{b}}\}$$

min : $f_{obj} = \left(\sum_{e=1}^{N_{e}} \rho_{e} V_{e}\right) / V_{0}$
s.t. $\begin{cases} \mathbf{K}\left(\overline{\rho}^{(i)}\right) \mathbf{U}^{(i)} = \mathbf{F} \\ g = \max_{i=1,\cdots,m} \left(c^{(i)}\right) - c_{limit} \leq 0 \\ \mathbf{z}_{min} \leq \mathbf{z} \leq \mathbf{z}_{max} \end{cases}$
(7)

where **z** is a set of design variable vectors, where each \mathbf{z}_k are up of 5 parameters, i.e., $\mathbf{z}_k = \{x_{k1}, y_{k1}, x_{k2}, y_{k2}, r_k\}$. ρ_e is the element densities defined in Eq. (6). \mathbf{z}_{\min} and \mathbf{z}_{\max} are the lower and upper bounds of the geometrical design variables. The volume fraction $f_{obj} = (\sum_{e=1}^{N_e} \rho_e V_e) / V_0$ is the objective function, where V_e and V_0 represent the volume of the *e*th element and design domain. N_e denotes the total number of elements. **F** is the global load vector. $c^{(i)} = \mathbf{F}^{\mathsf{T}}\mathbf{U}^{(i)}$, $\mathbf{K}(\overline{\rho}^{(i)})$ are the compliance and global stiffness matrix related to the (*i*)-th patch removal scenario. $\mathbf{U}^{(i)}$ can be obtained by solving the corresponding discrete equation $\mathbf{K}(\overline{\rho}^{(i)}) \mathbf{U}^{(i)} = \mathbf{F}$ for the (*i*)th patch removal scenario.

Since the maximum function in the constraint function of Eq. (7) is not differentiable, to make sure the design variables are updated by using a gradient-based algorithm, the non-differentiable max-operator max $(c^{(i)})$ is approximated by using a p-norm as

$$c^{PN} = c_{\max} \cdot \left(\sum_{i=1}^{m} \left(\frac{c^{(i)}}{c_{\max}}\right)^{q}\right)^{1/q}$$
(8)

where $c_{\text{max}} = \max(c^{(i)})$. q refers to the rate parameter, which determines how close c^{PN} is to the largest c_{max} in all patch removal scenarios, and sets as 10 for all examples in this paper. Theoretically, c^{PN} tends to approach c_{max} as the value of the parameter q approaches infinity. It should be noted that the parameter q should not be too large, a large value of q will lead to the instability of the optimization program. In this case, there is always a gap between the approximate value c^{PN} and the true maximum value c_{max} . In order to ensure the compliance value of the worst failure cases better meets the quantified design requirements defined by the user, the constraint function of Eq. (7) is replaced by the following equation:

$$\overline{g} = cp \cdot c^{PN} - c_{limit} \le 0 \tag{9}$$

where the correction parameter *cp* of the *n*th iteration is calculated by

$$cp_n = \frac{c_{\max,n}}{c_n^{PN}} \tag{10}$$

where $c_{\max,n}$ represents the compliance value corresponding to the worst failure case in the *n*th iteration.

3 RBTO of Fail-Safe Structures Considering Uncertainty

3.1 RBTO Model for Fail-Safe Structures Considering Uncertainty

All the related works on the fail-safe design of continua are undertaken under deterministic settings, disregarding the impact of uncertain factors on optimal design. However, uncertainties in structural parameters such as material property are inherent and the scatter from their nominal ideal values is unavoidable in the vast majority of engineering applications. A design derived through a

deterministic optimization process cannot guarantee the fulfilment of the constraints if some degree of uncertainty appears later in the problem. Consequently, these uncertainties should be accounted for during the structural design phase to reduce the risk of performance degradation. For the first time, we integrate reliability analysis considering uncertainty into FSTO. The RBTO model is formulated as

$$\begin{aligned} &\text{find} : \mathbf{z} = \left\{ \mathbf{z}_{1}, \mathbf{z}_{2}, \cdots, \mathbf{z}_{N_{b}} \right\} \\ &\text{min} : f_{obj} = \left(\sum_{e=1}^{N_{e}} \rho_{e} V_{e} \right) / V_{0} \\ &\text{s.t.} \begin{cases} \mathbf{K} \left(\overline{\rho}^{(i)}, \mathbf{X} \right) \mathbf{U}^{(i)} \left(\overline{\rho}^{(i)}, \mathbf{X} \right) = \mathbf{F} \left(\mathbf{X} \right) \\ &R = \Pr \left(\overline{g} \left(\mathbf{z}, \mathbf{X} \right) \le 0 \right) \ge \Phi \left(\beta^{t} \right) = R^{t} \\ &\mathbf{z}_{\min} \le \mathbf{z} \le \mathbf{z}_{\max} \end{aligned}$$
(11)

The main difference between RBTO of fail-safe structures in Eq. (11) and the classical DTO in Eq. (7) lies in the introduction of random parameter X (Namely, material parameter E_0 in Eq. (5), or the magnitude or direction of the load) that influence the structural responses. This induces the emergence of reliability constraint, which states that the probability of meeting the design requirements must not be less than the probability corresponding to the target reliability. Here the function \overline{g} is defined by Eq. (9), whose positive sign indicates violating a given constraint, i.e., when $\overline{g} > 0$ indicates a "fail" design. *R* is the reliability level, and *R'* is the target reliability level defined as the standard normal distribution $\Phi(\beta')$ at the target reliability index β' . All other parameters are identical as these in Eq. (7).

3.2 PMA-Based Reliability Analysis

Currently, numerous effective numerical reliability methods have been developed to approximate the reliability involved in reliability constraint, such as sampling-based methods utilizing Monte Carlo simulation and approximate reliability methods based on Taylor series expansion, etc. The reliability index approach (RIA) [44] and performance measure approach (PMA) [45,46] are two most commonly used first-order reliability method. Without loss of generality, here we use PMA to incorporate probability constraint in reliability analysis due to its stability and efficiency. Therefore, the PMA-based RBTO formulation is expressed mathematically as follows:

find:
$$\mathbf{z} = \{\mathbf{z}_1, \mathbf{z}_2, \cdots, \mathbf{z}_{N_b}\}$$

min: $f_{obj} = \left(\sum_{e=1}^{N_e} \rho_e V_e\right) / V_0$
s.t. $\begin{cases} \mathbf{K} \left(\overline{\rho}^{(i)}, \mathbf{X}\right) \mathbf{U}^{(i)} \left(\overline{\rho}^{(i)}, \mathbf{X}\right) = \mathbf{F} \left(\mathbf{X}\right) \\ \overline{g}^{PMA} \leq 0 \\ \mathbf{z}_{min} \leq \mathbf{z} \leq \mathbf{z}_{max} \end{cases}$
(12)

where \overline{g}^{PMA} denotes the performance measure associated with the target reliability β' , being evaluated via solving the following optimization problem:

$$\overline{g}^{\text{PMA}} = \arg\max_{\overline{g}} \left\{ \overline{g} \left(\mathbf{d}, \mathbf{Y} \right) | ||\mathbf{X}|| = \beta^{t} \left(= \Phi^{-1} \left(P^{t} \right) \right) \right\}$$
(13)

where Φ^{-1} refers to the inverse cumulative distribution function (CDF) of standard normal distribution.

In the PMA approach, a point satisfies this metric $||\mathbf{Y}|| = \beta^{t}$ and makes \overline{g} the maximum value is the most probable failure point (MPP) \mathbf{X}^{*} . Then, \overline{g}^{PMA} is calculated as $\overline{g}^{PMA} = \overline{g}(\mathbf{d}, \mathbf{Y}^{*})$ [47]. Among the several existing methods proposed to search the MPP, we choose to use the advanced mean value (AMV) method without loss of generality. The AMV formulation for MPP search is stated as [48]

$$\mathbf{Y}^{(k+1)} = \beta^{t} \mathbf{n}^{(k)} = \beta^{t} \frac{\nabla_{\mathbf{Y}} \overline{g} \left(\mathbf{Y}^{(k)} \right)}{||\nabla_{\mathbf{Y}} \overline{g} \left(\mathbf{Y}^{(k)} \right)||}$$
(14)

where **n** represents the steepest descending direction of the iteration point.

4 Design Sensitivity Analysis

4.1 Sensitivity of Performance Function w.r.t. Random Variables

Firstly, we focus on the sensitivity of the \overline{g} w.r.t. random material parameter (Young's modulus E_0). which can be calculated according to Eq. (9)

$$\frac{\partial \overline{g}}{\partial E_0} = cp_n \cdot \frac{\partial c^{PN}}{\partial E_0} = cp_n \cdot \sum_{i=1}^m \left(\frac{\partial c^{PN}}{\partial c^{(i)}} \frac{\partial c^{(i)}}{\partial E_0} \right)$$
(15)

In Eq. (15) the derivative term $\partial c^{PN} / \partial c^{(i)}$ can be derived from Eq. (8) as

$$\frac{\partial c^{PN}}{\partial c^{(i)}} = \left(\sum_{i=1}^{m} \left(\frac{c^{(i)}}{c_{\max}}\right)^{q}\right)^{1/q-1} \left(\frac{c^{(i)}}{c_{\max}}\right)^{q-1} \tag{16}$$

The adjoint variable method is used to calculate the derivative of the compliance $c^{(i)}$ of *i*-th patch removal scenario w.r.t. E_0 in Eq. (15). The augmented Lagrangian function constructs as

$$L^{(i)} = c^{(i)} + \lambda_{(i)}^{\mathrm{T}} \left(\mathbf{F} - \mathbf{K}^{(i)} \mathbf{U}^{(i)} \right) = \mathbf{F}^{\mathrm{T}} \mathbf{U}^{(i)} + \lambda_{(i)}^{\mathrm{T}} \left(\mathbf{F} - \mathbf{K}^{(i)} \mathbf{U}^{(i)} \right)$$
(17)

where $\lambda_{(i)}$ refers to the adjoint vector corresponding to *i*th patch removal scenario.

Then the $\partial \mathbf{L}^{(i)} / \partial E_0$ gives as

$$\frac{\partial L^{(i)}}{\partial E_0} = -\lambda_{(i)}^{\mathrm{T}} \frac{\partial \mathbf{K}^{(i)}}{\partial E_0} \mathbf{U}^{(i)} + \left(\mathbf{F} - \lambda_{(i)}^{\mathrm{T}} \mathbf{K}^{(i)}\right) \frac{\partial \mathbf{U}^{(i)}}{\partial E_0}$$
(18)

To eliminate $\partial \mathbf{U}^{(i)}/\partial E_0$, let the vector $\lambda_{(i)}$ in Eq. (18) be the solution of equation $\mathbf{K}^{(i)}\lambda_{(i)} = \mathbf{F}$, bringing about $\lambda_{(i)} = \mathbf{U}^{(i)}$. Therefore, the derivative $\partial c^{(i)}/\partial E_0$ in Eq. (15) is explicitly determined as

$$\frac{\partial c^{(i)}}{\partial E_0} = \frac{\partial L^{(i)}}{\partial E_0} = -\left(\mathbf{U}^{(i)}\right)^{\mathrm{T}} \frac{\partial \mathbf{K}^{(i)}}{\partial E_0} \mathbf{U}^{(i)} = \sum_{e=1}^{N_e} \left[-\left(1 - E_{\min}\right) \left(\rho_e^{(i)}\right)^p \left(\mathbf{u}_e^{(i)}\right)^{\mathrm{T}} \mathbf{k}_0 \mathbf{u}_e^{(i)} \right]$$
(19)

The derivative of performance function w.r.t. the random load can be derived similarly, and the details are omitted here.

4.2 Sensitivity of Performance Function w.r.t. Design Variables

Now, we focus on the sensitivity of \overline{g} w.r.t. design variables. Assuming ψ_k represents one of the geometric design variables $\mathbf{z}_k = \{x_{k1}, y_{k1}, x_{k2}, y_{k2}, r_k\}$ for the *k*th moving bar, the design sensitivities $\partial \overline{g} / \partial \psi_k$ can be calculated as

$$\frac{\partial \overline{g}}{\partial \psi_k} = cp_n \cdot \frac{\partial c^{PN}}{\partial \psi_k} = cp_n \cdot \sum_{i=1}^m \left(\frac{\partial c^{PN}}{\partial c^{(i)}} \frac{\partial c^{(i)}}{\partial \psi_k} \right)$$
(20)

where the derivative term $\partial c^{PN} / \partial c^{(i)}$ is determined by Eq. (16). The derivative term $\partial c^{(i)} / \partial \psi_k$ can be determined by the chain rule as

$$\frac{\partial c^{(i)}}{\partial \psi_k} = \sum_{e=1}^{N_e} \left(\frac{\partial c^{(i)}}{\partial \overline{\rho}_e^{(i)}} \frac{\partial \overline{\rho}_e^{(i)}}{\partial d_{ek}} \frac{\partial d_{ek}}{\partial \psi_k} + \frac{\partial c^{(i)}}{\partial \overline{\rho}_e^{(i)}} \frac{\partial \overline{\rho}_e^{(i)}}{\partial r_k} \frac{\partial \overline{\rho}_e^{(i)}}{\partial \psi_k} \frac{\partial r_k}{\partial \psi_k} \right)$$
(21)

Similar to the calculation process of $\partial c^{(i)}/\partial E_0$, we can also use the adjoint variable method to derive the derivative term $\partial c^{(i)}/\partial \rho_e^{(i)}$ in Eq. (21)

$$\frac{\partial \boldsymbol{c}^{(i)}}{\partial \overline{\rho}_{e}^{(i)}} = -\frac{\partial \boldsymbol{E}_{e}}{\partial \rho_{e}^{(i)}} \left(\mathbf{u}_{e}^{(i)} \right)^{\mathrm{T}} \mathbf{k}_{0} \mathbf{u}_{e}^{(i)} = -p \left(\boldsymbol{E}_{0} - \boldsymbol{E}_{\min} \right) \left(\rho_{e}^{(i)} \right)^{p-1} \left(\mathbf{u}_{e}^{(i)} \right)^{\mathrm{T}} \mathbf{k}_{0} \mathbf{u}_{e}^{(i)}$$
(22)

In Eq. (21), $\partial \rho_e^{(i)} / \partial d_{ek}$ and $\partial \rho_e^{(i)} / \partial r_k$ can be determined according to Eq. (1), gives

$$\frac{\partial \rho_e^{(i)}}{\partial d_{ek}} = -\left(1 - \rho_e^{(i)}\right) \frac{\beta \cdot \exp\left[-\beta (d_{ek} - r_k)\right]}{1 + \exp\left[-\beta (d_{ek} - r_k)\right]}, \frac{\partial \rho_e^{(i)}}{\partial r_k} = -\frac{\partial \rho_e^{(i)}}{\partial d_{ek}}$$
(23)

In Eq. (21), $\partial d_{ek}/\partial \psi_k$ and $\partial r_k/\partial \psi_k$ can be determined from Eqs. (1)–(4). For more details see [39]. The sensitivity information of the objective function can also be obtained similarly and ignored here. Once the sensitivity information is obtained, the well-known method of moving asymptotes (MMA) developed by Svanberg [49] can be used to update design variables. The iterative process terminates when a given maximum number of iterations is achieved.

5 Numerical Examples

The proposed method for RBTO of fail-safe design under uncertainty is demonstrated in three numerical examples in this section. Unless otherwise stated, E_0 is regarded as a probabilistic random variable with normal distribution, and its mean and coefficient of variation (COV) are assumed to be 2×10^5 and 0.1. Poisson's ratio is fixed at 0.3.

5.1 Clamped Beam

First, a design problem of clamped beam is tested. Fig. 4a depicts the design domain with dimensions of 200×100 and boundary conditions. Three quarters of the left and right sides of clamped beam are completely constrained and the center of the upper edge is subjected to a concentrated force of 1000. A damage population consisting of 8 square damage areas with a side length of 50 is considered and shown in Fig. 4a, where labels (1–8) are marked for the damage zones. 200×100 plane stress elements of side-length 1 are used to discretize the design domain. The initial layout of 208 MMBs with thickness $t_0 = 2r_0 = 4.5$ is shown in Fig. 4b. The compliance limit value c_{limit} is set to be 130.

This problem is first designed for standard deterministic design without considering local damage, marked as Case 1. The final topology aims at minimizing the structure's volume fraction with compliance constraint, as shown in Fig. 5a, where Young's modulus is set as 2×10^5 for DTO design. Second, this problem is also designed for deterministic fail-safe design based on the optimization model defined in Eq. (11), marked as Case 2. The final topology of DTO design considering local damage is shown in Fig. 5b. To investigate the influences of uncertainty material parameter (Young's modulus E_0) on optimization results of fail-safe structures, we revisit the clamped beam problem for reliability design based on the RBTO method described in Section 3, marked as Case 3. For RBTO

design, the target reliability index in Eq. (11) is set as 1 in this example. The final topology of RBTO design considering local damage is shown in Fig. 5c. To have a more direct observation of how the structure maintains some degree of integrity under each independent failure mode, the final results of DTO and RBTO design considering local damage are depicted in Figs. 6 and 7, respectively.



Figure 4: Clamped beam problem: (a) design domain and boundary condition, (b) initial layout of 208 bars with thickness of $t_0 = 4.5$



Figure 5: Final topology results of clamped beam: (a) DTO design without considering local damage, (b) DTO design considering local damage, (c) RBTO design considering local damage



Figure 6: Damaged models with final fail-safe topology for DTO design

When comparing Fig. 5a with Figs. 5b and 5c, a significant topological difference can be observed between the standard design (Fig. 5a) and fail-safe designs (Figs. 5b and 5c). Compared with standard design, more load path redundancy occurs in fail-safe designs. Similarly, more redundancy joints and smaller features can be observed in RBTO design considering local damage compared to DTO design considering local damage. The topological differences between these three cases emphasize the necessity and importance of integrating fail-safe concept and reliability analysis under material uncertainty into TO.



Figure 7: Damaged models with final fail-safe topology for RBTO design

In Table 1 the volume fractions of the DTO designs and RBTO design for clamped beam are listed, as well as the compliances for all eight damage instances. The values of compliance in three different cases are calculated based on the final topology shown in Fig. 5. For example, the eight compliance values of Case 1 are for the DTO design (in Fig. 5a) under eight damage states. The difference between Case 3a and Case 3b is that the compliance values for Case 3a are calculated based on the most probable failure point of the random variable (namely, the value of Young's modulus included in the calculation in this case is 1.8×10^5), while the compliance values for Case 3b are calculated based on the mean value of the random variable (namely, Young's modulus included in the calculation is 2×10^5).

Cases	Volume fraction		Failure zones						
		1	2	3	4	5	6	7	8
Case 1	0.091	415.24	1.047×10^{5}	1.0512×10^{5}	417.51	1.0468×10^{5}	130	130	1.0511×10^{5}
Case 2	0.362	89.77	129.89	130.00	89.576	101.10	105.31	104.90	100.96
Case 3a	0.424	89.652	129.94	130.00	89.61	100.07	102.96	102.81	100.22
Case 3b	/	80.687	116.95	117.00	80.649	90.06	92.664	92.531	90.199

Table 1: Optimization results of volume fractions and compliances for clamped beam

Table 1 shows that the relative volume fractions are 0.091 for Case 1, 0.362 for Case 2, and 0.424 for Case 3, respectively. The final compliance values (or compliance of the worst failure cases) satisfy the constraint threshold of 130 well for three different cases. For DTO design without considering fail-safe requirement in Case 1 (Fig. 5a), once the materials in failure zones of the final design are removed, the corresponding compliance values far exceed the constraint threshold of 130, except for the 6th and 7th failure zones. There is no doubt that this DTO design will be extremely sensitive to local failure of material in the 1st–5th and 8th failure zones due to its lack of redundancy. At this point, the structure may fail when the material in these zones is removed resulting from a partial collapse. For fail-safe designs in Case 2 (Fig. 5b) and Case 3a (Fig. 5c), the compliances for all eight failure modes are all less than or equal to the constraint threshold of 130. This can be easily understood as we constrain the compliance value of the worst failure cases. From another point of view, it is proved that increasing structure redundancy is conducive to improving the resistance of the structure to local failure. Fail-safe designs can maintain certain service performance even if the local material is removed or the local stiffness is lost, which can be clearly seen from Figs. 6 and 7. Besides, we can find that the 2nd and 3rd failure zones are the core areas of fail-safe designs in Figs. 5b and 5c, which can be seen

as the worst failure cases in this example. Because the compliance values (the numbers in bold type in Table 1) corresponding to the 2nd and 3rd patch removal scenarios directly approximate the constraint threshold of 130. In Case 3b the compliance values for RBTO design considering local damage are less than that of DTO design for all eight failure modes. We can conclude that integrating reliability criterion into fail-safe design further increases the robustness of the designs towards local failure.

To show the influence of different shapes and arrangements of failure regions on the optimization results, we solve the above clamped beam example with 16 damage areas. The final topologies of RBTO designs considering local damage are depicted in Table 2. Significant topological differences can be observed for optimized structures with shapes and arrangements of failure regions, compared to the final topology in Fig. 5c. Therefore, we need to establish a relatively appropriate failure block model according to the possible failure area for practical problems.



Table 2: Optimization results for different shape and arrangement of failure region

To show the superiority and computational effectiveness of the proposed method, we revisit the above clamped beam example considering $\beta^i = 1$ using the well-known SIMP method [50]. The design domain is discretized with 200 × 100 elements, which means that there will be 200 × 100 density design variables in the SIMP method. However, the present method has only 208 × 5 = 1040 geometric design variables. Fig. 8 shows the topology optimization results of the clamped beam based on the simp method. Table 3 lists the average computation time (in seconds) of the SIMP and MMB methods, where the computation times are calculated as the average of the entire optimization process (250 iterative steps). The average time of finite element analysis of two methods is about the same since the same number of finite elements is used for the structural analysis. However, the average computation times of using MMA algorithm to update design variables and one optimization step in the MMB method are smaller than the corresponding times in the SIMP method. Moreover, the Monte Carlo simulation (MCS) with 10000 sample points is performed, and the reliability index results of MCS are

listed in Table 3. It is seen that the final structure can meet the requirements of target reliability $\beta^t = 1$ well, which means that the proposed method is effective.



Table 3:	А	com	parison	of	com	putation	time	for	clam	ped	beam

Method	Volume fractions	Reliability index of MCS		CPU time (s)		
			FEA	MMA	One optimization step	
SIMP	0.405	1.0102	1.715	1.088	8.101	
MBB	0.424	1.0165	1.740	0.037	7.732	

5.2 Cantilever Beam

Second, the classical design problem of cantilever beam subjected to the bending force used in [3,4] is considered. Fig. 9 shows the design domain with dimensions of 180×60 , as well as the initial layout of 208 MMBs of thickness $t_0 = 4.5$ was used for designing the topology of the load transfer path. The left side of the cantilever beam is completely fixed and the center point of the right side is subjected to a vertical concentrated force of 1000. A damage population consisting of 12 square damage areas with a side length of 30 is considered and shown in Fig. 9a, where labels (1–12) are marked for the damage zones. 180×60 plane stress elements of side-length 1 are used to discretize the design domain. The compliance limit value c_{limit} is set as 5000.



Figure 9: Cantilever beam problem: (a) design domain and boundary condition, (b) initial layout of 208 bars with thickness of $t_0 = 4.5$

Fig. 10 investigates the effect of reliability indexes on the FSTO by setting ($\beta^t = 2, 3, 4$). The DTO model considering the fail-safe requirement in Eq. (11) is also performed for comparison and the result is shown in Fig. 10d. Observably, the final topologies for RBTO designs and DTO designs differ significantly, highlighting the significance of integrating material uncertainty into fail-safe designs.





Besides, the difference between the RBTO designs in Figs. 10a–10c indicates that the reliability indexes have a greater impact on the topological layout of fail-safe designs.

Figure 10: Optimization results of cantilever beam without minimum size control: RBTO designs in (a–c) and DTO design in (d)

In Table 4 the volume fraction and minimum size of the deterministic fail-safe design and reliability-based fail-safe designs for the cantilever beam are listed, as well as the compliances for the selected six failure modes. Due to the symmetry of the structure, only the 1th–6th failure modes are selected here. From Table 4, we can see that the volume fraction is 0.388 for deterministic fail-safe design in Fig. 10d, and 0.487, 0.554, 0.671 for reliability-based fail-safe designs with $\beta' = 2, 3, 4$ in Figs. 10a–10c, respectively. The minimum size is 0.585 for the DTO design and 0.749, 0.598, 1.017 for RBTO designs with $\beta' = 2, 3, 4$. We can conclude that improving of structural reliability of fail-safe designs usually requires more material consumption. We can also observe that the compliance values of the worst failure cases for deterministic and reliability-based fail-safe designs satisfy the constraint threshold of 5000 wells. For DTO design and RBTO designs with $\beta' = 2$ and 3, the 2nd and 8th failure zones are the core areas of fail-safe designs, which correspond to the worst failure cases. The compliance values corresponding to the 4th–6th patch removal scenarios for RBTO designs are less than those of the DTO design for these patch removal scenarios.

Cases	Volume fraction Minimum size min $(2r_k)$		Selected failure zones						
			1	2	3	4	5	6	
$\beta^t = 2$	0.487	0.749	4870.5	5000.0	4530.2	3882.5	3668.5	3390.3	
$\beta^t = 3$	0.554	0.598	4868.1	5000.1	4427.9	3768.2	3628.4	3532.2	
$\beta^t = 4$	0.671	1.017	5000.0	4985.0	4191.6	3696.2	3369.6	3320.3	
DTO	0.388	0.585	4878.8	5000.0	4533.5	3942.8	3640.6	3511.3	

 Table 4: Optimization results of cantilever beam without minimum size control

Next, we revisit the fail-safe design problem of cantilever beam with a minimum length scale to demonstrate the advantages of using the MMB method for fail-safe designs. It should be stressed that imposing a minimum size control is easy to achieve by simply assigning a lower bound r_{min} to the

design variable r_k in the proposed method. This implementation can improve the manufacturability of optimized structures.

Fig. 11 shows the resulting topologies for deterministic and reliability-based fail-safe designs with an imposing minimum length scale $2r_{min} = 2$. Apparent differences can be observed for the final topologies without minimum size control in Figs. 10a–10d and those with minimum size control in Figs. 11a–11d. The volume fraction and minimum size for DTO and RBTO designs are listed in Table 5, as well as the compliances for the selected six failure modes. We can see that the minimum sizes of the final designs and the compliance values of the worst failure cases satisfy the constraint limits well for different target reliability indexes. Another noteworthy phenomenon is that the compliance values corresponding to the 4th–6th patch removal scenarios for RBTO designs with minimum size control are less than those without minimum size control when the same level of reliability is met. This means that reliability-based fail-safe designs with minimum size control are less sensitive to these possible failure zones compared to the cases without minimum size control for the cantilever beam.



Figure 11: Optimization results of cantilever beam with minimum size control: RBTO designs in (a-c) and DTO design in (d)

Cases	Volume fraction Minimum size min $(2r_k)$		Selected failure zones							
			1	2	3	4	5	6		
$\beta^t = 2$	0.488	2.00	4881.9	4999.9	4512.7	3824.6	3600.7	3229.9		
$\beta^t = 3$	0.565	2.00	4869.2	4999.2	4407.6	3773.9	3526.1	3241.2		
$\beta^t = 4$	0.672	2.00	5000.0	4990.8	4210.9	3565.9	3389.7	3031.3		
DTO	0.399	2.00	4913.2	4999.7	4501.7	3899.1	3923.6	3443.3		

 Table 5: Optimization results of cantilever beam with minimum size control

5.3 L-Shaped Beam

Finally, an L-shaped beam problem sketched in Fig. 12a is tested. The initial layout of 232 MMBs of thickness $t_0 = 2r_0 = 5$ is shown in Fig. 12b. The top side of the L-shaped beam is completely fixed, and a concentrated force of F = 1000 is acted on the top right. 25,600 plane stress elements of side-length 1 are used to discretize the design domain. The compliance limit value c_{limit} is set as 6000 in this example.



Figure 12: L-shaped beam problem: (a) design domain and boundary condition, (b) initial layout of 232 bars with thickness of $t_0 = 5$

The results of the fail-safe design for RBTO with various reliability indexes and DTO are compared. A minimum size control is imposed by assigning a lower bound $r_{\min} = 1$ for RBTO and DTO designs. The RBTO results for $\beta' = 2, 3, 4$ are presented in Fig. 13, as well as the DTO result obtained based on Eq. (11). The volume fractions of deterministic and reliability-based fail-safe designs for L-shaped beam are given in Table 6, as well as the compliances for 14 different failure modes. We can also observe that the compliance values of the worst failure cases for deterministic and reliability-based fail-safe designs satisfy the constraint threshold of 6000 well. The phenomenon that the volume fraction increases with increasing reliability requirements indicates more materials are required to ensure the safety of fail-safe structures.



Figure 13: (Continued)



Figure 13: RBTO designs for different target reliability indexes (a-c), and deterministic design (d)

Cases	Volume fraction		ies					
		1	2	3	4	5	6	7
$\beta^t = 2$	0.409	4347.6	4970.7	5218.7	5747.3	4987.7	6000.1	3892.6
$\beta^{t} = 3$	0.468	4100.1	4934.4	5052.4	5770.0	4921.4	6000.1	3796.6
$\beta^t = 4$	0.560	3957.5	4914.8	4911.8	5734.3	4623.3	6000.0	3532.2
DTO	0.340	4490.3	4998.5	5280.2	5664.2	5023.9	6000.0	3966.5
		8	9	10	11	12	13	14
$\beta^t = 2$		3570.9	3763.1	4470.8	5138.1	4643.7	4453.6	3900.8
$\beta^t = 3$		3534.3	3164.1	4267.4	4968.1	4518.6	4201.9	3984.5
$\beta^{t} = 4$		3758.3	2709.7	3955.9	4840.2	4423.5	4248.7	3906.9
DTO		3596.9	4008.3	4457.5	5199.0	4677.9	4324.4	4202.0

Table 6: Optimization results of volume fractions and compliances for L-shaped beam

Fig. 14 shows the iterative histories of volume fractions of the L-shaped beam. As anticipated, the convergence of volume fraction stability demonstrates the proposed method operated very well in RBTO of fail-safe structures under material uncertainty.



Figure 14: Iterative histories of volume fractions of L-shaped beam

The present method can also be extended to solve the optimization problem considering load uncertainty. We revisit the above L-shaped beam example considering the uncertainty of load magnitude, where the load is assumed as a normally distributed random variable with a mean of 1000 and a standard deviation of 50. The RBTO results for $\beta^t = 1, 2, 3$ are presented in Fig. 15, as well as the DTO result obtained based on Eq. (11). Moreover, Monte Carlo simulation (MCS) with 10000 sample points is also performed to verify the effectiveness of the proposed method. The optimization results are listed in Table 7. It is seen that the reliability indexes obtained based on the present method are in good agreement with the Monte Carlo simulation, and the relative error of the reliability index of the optimized structure does not exceed 3%, which also proves that the present method also performs well in an optimization problem considering load uncertainty.



Figure 15: (Continued)



Figure 15: RBTO designs for different target reliability indexes (a-c), and deterministic design (d)

Cases	Volume fractions	Reliability index of MCS
$\beta^t = 1$	0.362	1.0135
$\beta^t = 2$	0.397	1.9972
$\beta^t = 3$	0.436	2.9678
DTO	0.332	/

Table 7: Reliability index of MCS for L-shaped beam

6 Conclusions

This paper developed a reliable RBTO methodology for solving the design problem of failsafe structures considering uncertainty based on the explicit moving morphable bars method, which integrates the PMA-based reliability analysis into the free-form design of fail-safe structures. A differentiable p-norm performance function with a correction parameter is introduced to ensure the compliance value of the worst failure case accurately meets the design requirement. Numerical examples emphasize the necessity and significance of incorporating reliability analysis under uncertainty into fail-safe systems and demonstrate that the reliability-based fail-safe designs are safer than the deterministic fail-safe design, and the incorporation of reliability criteria into the fail-safe design can further improve the designs' resistance to local failure. In addition, the notable advantage of computational efficiency and realizing minimum constraint control using the MMB method to implement topology optimization is also well-reflected.

Acknowledgement: The authors are grateful to Prof. Krister Svanberg for providing the procedure of MMA algorithm.

Funding Statement: This work is supported by the National Natural Science Foundation of China (Grant No. 12172114), Natural Science Foundation of Anhui Province (Grant No. 2008085QA21), Fundamental Research Funds for the Central Universities (Grant No. JZ2022HGTB0291) and China Postdoctoral Science Foundation (Grant No. 2022M712358).

Conflicts of Interest: The authors declare that they have no conflicts of interest to report regarding the present study.

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